Sensor fusion for motion processing and visualization

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What we have - Shimmer Wireless Sensor Platform

Sensors

- 3-axis accelerometer, Freescale MMA7260Q ±1.5/2/4/6g (1g≈9.81m/s²)
- 3-axis gyroscope; two integrated dual-axis angular rate gyroscopes InvenSense 500 series
- No compass

Processing

- MSP430[™]16-bit Ultra-Low Power MCU @ 8 MHz
- 10Kbyte RAM, 48Kbyte ROM
- 8 Channels of 12bit A/D

Battery

Integrated Li-ion
 280 mAh, 3.7 V





What we have - Shimmer Wireless Sensor Platform (continued)

Radios

- 2.4 GHz IEEE 802.15.4 Chipcon CC2420
- Mitsumi WML-C46N CSR based Class 2 Bluetooth Radio

Storage

2 GB Micro SD card

Form factor

- Small form factor
- 50mm x 25mm x 12.5mm
- Light weight: 15 grams

Software

- TinyOS event driven OS for WSN
- Open source



What we want - Gait analysis

- Analysis of measurable parameters of human gait
- From the output of the sensors -> reconstruct the orientation of the limbs in time
- •The orientation, described by angles and position



Showing live demo

Qt cross-platform GUI, C++ programming language

Open-source

For medical use (research)

Elbow flexion (left and right arm)

Results are presented as animation (OpenGL)



What is the competitive edge of the Shimmer platform?

Competitive if:

- Video cameras / ultrasound measuring systems are not applicable; for example outdoor activities
- Multiple sensors are needed
- We cannot or we do not want to process the data real-time

Kinect Guiness record, 8m in 2months

XBOX 360



Smart phones, Android devices

So how can we compute it?



From the output of the sensors -> reconstruct the orientation of the limbs in time



What does the output of the sensors look like?



•The sensor was placed on the desktop then turned over

• Output is not zero even in the static case

• Axes are not necessarily perpendicular

 Needs calibration,
 linear transfer function is assumed

• Accelerometer measures gravity even if the mote is stationary

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•The sensor was placed on the desktop then turned over

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• Accelerometer measures gravity even if the mote is stationary In the static case constant 1g pointing downwards is measured

The angle between an axis of the device and the horizontal plane can be computed – **Tilt angles**



We could even compute orientation if we had a compass (accelerometers have no idea where north is)

Tilt angles computed from the acceleration data

Tilt angles computed from the measured acceleration (alpha) practically coincides with the true tilt angles (beta) if the mote is not accelerating We get weird **spikes** when the sensor is accelerating



Workaround: low-pass filter, works but poor transient response, lags

Gyroscopes measures angular rate in the **mote frame** of reference

Angles can be computed by numerical integration:

$$\theta(t) = \theta(0) + \int_{0}^{t} \omega(\tau) d\tau$$
 (simplified!)

Good news: not influenced by acceleration or gravity; fast, responsive, not subject to lag **Orientation s.t. initial orientation can be computed**

Drift

The integrated effects over time of a slowly varying offset and noise. The drift must be

eliminated, requires an external reference vector that does not drift.





Putting it all together – drift cancellation

Both accelerometer and gyro data can be used to **compute angles**

Accelerometer provides a reference vector, gravity, that does not drift (long-term) but has poor transient response

Gyro data is excellent for computing orientation (short-term), gives fast transient response but needs reference vector that does not drift



Our approach: offline, nonlinear regression for each record Custom C++ solver using IPOPT

Future plans

Appropriate uses:

- Outdoor activities
- Multiple sensors
- Off-line processing

Plans:

- Rowing
- Running



Prof. László Hatvani: technical discussion

Péter Ruzicska: colleague

Miklós Maróti: supervising the research

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Shane Colton; *The Balance Filter, A Simple Solution for Integrating Accelerometer and Gyroscope Measurements for a Balancing Platform*; Chief Delphi white paper, 2007. http://web.mit.edu/scolton/www/filter.pdf

Further slides, not presented

Live demo – fooling the application

- After drift cancellation, only the gyro data is used to compute orientation
- Position is not computed
- We can fool the application by rotating the sensor around a point
- It still gives us a proper elbow flexion



Sensors: accelerometer

Accelerometer

- 3 Axis Accelerometer, Freescale MMA7260Q
- Sensitivity: 800 mV/g @ 1.5g
- 12 bit analogue digital converter → integer number
 Resolution: (1.5g+1.5g)/(2¹²)≈7·10⁻⁴g/unit





Accelerometer measures gravity + acceleration

Comparing the measured acceleration (ax, ay, az) and the gravitational acceleration (bx, by, bz) in the mote frame Workaround: **low-pass filter**, works but poor transient response, **lag**



Static calibration of the accelerometer

Assumption: measured value is a linear function of the acceleration (linear transfer function)
 Calibration: find the gain matrix (9 unknowns) and offset vector (3 unknowns)

acceleration [m/s²] = gain·(measured value) – offset

Place the mote on each of its six side and record the output acceleration: $(\pm 1g, 0, 0)$; $(0, \pm 1g, 0)$; $(0, 0, \pm 1g)$



Gives an overdetermined system of linear equations (18 equations and 12 unknowns); linear least-squares, analytic solution (SVD)

Sensors: accelerometer and gyroscopes

Accelerometer

- 3 Axis Accelerometer, Freescale MMA7260Q
- Sensitivity: 800 mV/g @ 1.5g
- 12 bit analogue digital converter -> integer number
- Resolution: $(1.5g+1.5g)/(2^{12}) \approx 7 \cdot 10^{-4} \text{ g/unit}$

Gyroscope

- 2 integrated dual-axis, InvenSense 500 series
- Full scale range: +/- 5000 deg/s
- Sensitivity: 2 mV/deg/s

Output in the static case

- Offset (**not zero**), actual value depends mainly on chip, plus temperature, *etc.*
- Accelerometer: constant value corresponding to 1g (gravity of Earth)



Sensors: gyroscopes

Gyroscope

- 2 integrated dual-axis, InvenSense 500 series
- Measures angular rate
- Full scale range: +/- 5000 deg/s
- Sensitivity: 2 mV/deg/s
- 12 bit ADC -> integer number



Output in the static case

- Offset (**not zero**), actual value depends mainly on chip, plus temperature, *etc.*
- Accelerometer: constant value corresponding to 1g (gravity of Earth)

Calibration of the gyroscopes

Calibration: find the gain matrix (9 unknowns) and offset vector (3 unknowns; linear transfer function)

angular rate [rad/s] = gain (measured value) - offset

Place the mote on each of its six side and record the output angular rate: $(\pm 45$ rpm, 0, 0); $(0, \pm 45$ rpm, 0); $(0, 0, \pm 45$ rpm)



A small error in the offset accumulates -> huge error in orientation over time, **drift**

Can we compute speed or position from a(t)?

$$v(t) = v(0) + \int_{0}^{t} (a(\tau) - g) d\tau$$

$$r(t) = r(0) + \int_{0}^{t} v(\tau) d\tau$$

$$earth frame$$

$$r(t) = r(0) + \int_{0}^{t} v(\tau) d\tau$$

The a(t) vector is measured in the **mote frame** of reference but we would like to track the mote in the **earth frame**.

Transformation is needed from one frame to the other -> rotation

Rotation

We need rotation to transforms the acceleration vectors from the mote frame of reference to the earth reference

Rotation: linear transformation, preserves lengths of vectors and angles between vectors



Rotation matrix

Rotation (continued)

Rotation representations

- Rotation matrix
- Euler angles
- Angle/axis
- Quaternion



Rotation is uniquely defined by 3 angles

Storage requirements

- matrix 9
- quaternions 4
- angle/axis 3

Performance comparison of rotation chaining operations multiplies add/substr. total

matrix	27	18	45
quaternions	16	12	28

Performance comparison of vector rotating operations

	multiplies	add/substr.	sin/cos	total
matrix	9	6	0	15
quaternions	21	18	0	39
angle/axis	23	16	2	45

Rotation in 3D is generally not commutative (neither is matrix multiplication)

The order in which infinitesimal rotations are applied is irrelevant

Rotation matrix of infinitesimal rotations along the x, y, z axis:

$$\begin{bmatrix} 1 & -d\Theta_{z} & d\Theta_{y} \\ d\Theta_{z} & 1 & -d\Theta_{x} \\ -d\Theta_{y} & d\Theta_{x} & 1 \end{bmatrix}$$

Gives a recipe to update the rotation matrix from gyro signals

W. Premerlani and P. Bizard; *Direction Cosine Matrix IMU: Theory*

$$R(t+dt) = R(t) \begin{bmatrix} 1 & -d\Theta_z & d\Theta_y \\ d\Theta_z & 1 & -d\Theta_x \\ -d\Theta_y & d\Theta_x & 1 \end{bmatrix} \begin{bmatrix} d\Theta_x = \omega_x dt \\ d\Theta_y = \omega_y dt \\ d\Theta_z = \omega_z dt \end{bmatrix}$$

Sources of errors

- Finite time step
- Quantization error: finite digital representation

The rotation matrix must be corrected -> **renormalization** at each point (no divisions or square roots)

Drift cancellation off-line with nonlinear regression

Off-line, operates on the whole data set but manipulates only 12 variables: 3x3 gain matrix and offset vector of the gyro

angular rate [rad/s] = gain (measured value) - offset

Assumption: 'On average' the measured acceleration points into the same direction (gravitational acceleration).

A nonlinear programming problem is solved

$$\max \left| \sum_{i=0}^{N} R_{i} a_{i} \right|$$

$$R(0) = I$$

$$R(i) = R(i-1)G(i-1) \text{ for } i=1...N$$

$$G(i): \text{ from gyro signals}$$

variables: gyro gain and offset

Software

- NLP, a nonlinear programming problem to be solved
- **IPOPT**, general purpose NLP solver (line search filter method) remarkably robust
- C++ API is used, only the objective has to be implemented
- Automatic differentiation (AD): the gradient is <u>not</u> approximated with numerical differentiation but automagically computed with AD (our own C++ library)
- L-BFGS (approximates the inverse Hessian matrix) to further speed up the computations

Perhaps the most popular one is the Kalman Filter

It is an on-line method that manipulates each sample

Good introduction with examples: SIGGRAPH Course Pack

Kalman filter is difficult to understand

Much simpler approach is the Complementary Filter with similar results (see filter.pdf)

Computing position

Now, the measured values in the mote frame can be transformed to the earth frame



Horrible error, computing the double integral is notoriously difficult